

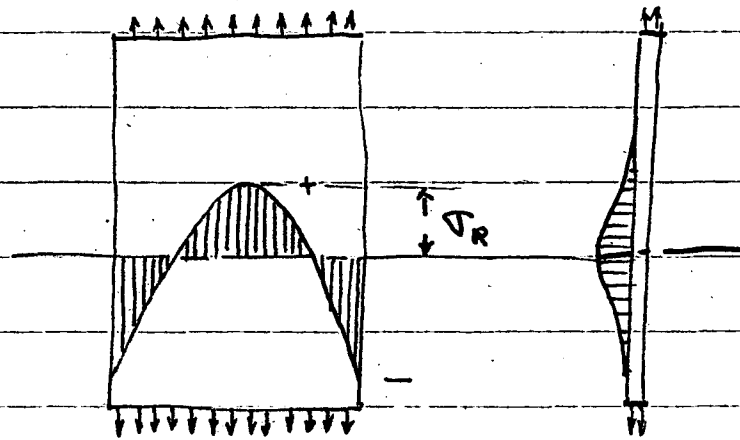
Effects of Residual Stress ~~to~~ on  
the ~~the~~ behaviors of Beams.

" Residual stress makes an  
elastic problem plastic "

~~April~~

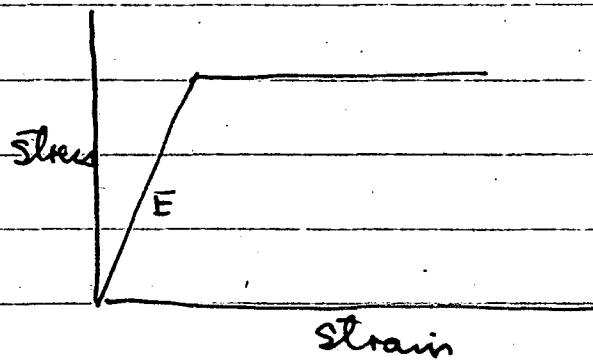
May 1, 1949

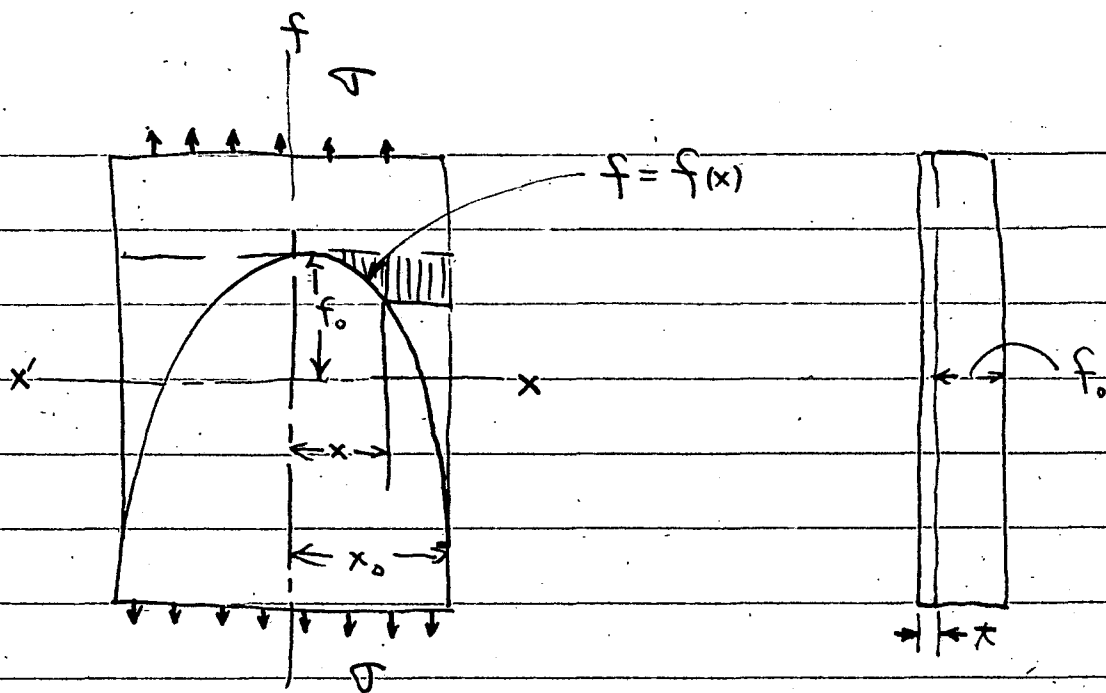
# Residual stress under simple tension.



Assumption:

The stress and strain relation for the material follows the following diagram





A.  $A = x_0 t$   
 $A_E = \text{Elastic area}$   
 $A_E = t(x_0 - x)$

$$E_p = E \frac{A_E}{A} = \frac{x_0 x}{x_0} = E \left(1 - \frac{x}{x_0}\right) \quad \text{--- (A)}$$

Assume  $f_0 = f_{y.p}$

$$\sigma A = \int_0^x t(f_0 - f) dx + t(x_0 - x)f$$

$$\sigma x_0 = \int_0^x (f_0 - f) dx + (x_0 - x)f$$

So we solve  $x = F(\sigma) \quad \text{--- (B)}$

Substitute (B) to (A)

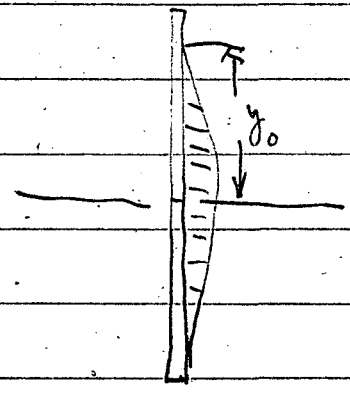
$$\therefore E_p = E \left(1 - \frac{F(\sigma)}{x_0}\right)$$

$$\bar{E}_p = \frac{d\sigma}{d\epsilon}$$

$$\therefore \epsilon = \int_0^{\sigma} \frac{d\sigma}{\bar{E}_p}$$

$$\Delta L = L \times \epsilon$$

$$\Delta L = L \int_0^{\sigma} \frac{d\sigma}{\bar{E}_p}$$



B. how if  $f_0$  vary with  $y$  axis

$$f_0 = f(y_0) \quad \text{but } y=0 \quad f(y_0) = f_{y.p.}$$

$$\therefore d(\Delta L) = \frac{L}{E} \int_0^{f_{y.p.} = f(y)} d\sigma \quad dy$$

$$+ \int_0^{f_{y.p.}} \frac{d\sigma}{\bar{E}_p} \quad dy$$

$$\Delta L = \int_0^{y_0} \left[ \frac{L}{E} \int_0^{f_{y.p.} = f(y)} d\sigma + \int_{f_{y.p.} = f(y)}^{f_{y.p.}} \frac{d\sigma}{\bar{E}_p} \right] dy$$

C. If the applied stress vary along the y axis.

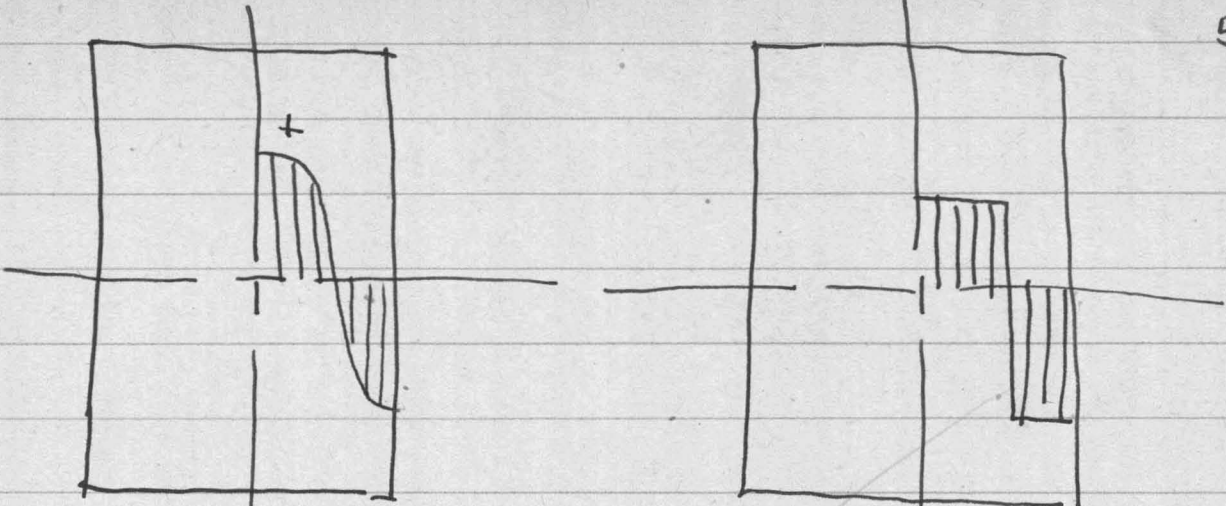
$$\Delta L = \int_0^{y_0} \left[ \frac{1}{E} \int_{f_{y.p.} - g(y)}^{f_{y.p.} + g(y)} d\sigma + \int_{f_{y.p.} - g(y)}^{Q(y)} \frac{d\sigma}{E_p} \right] dy$$

where  $Q(y) = \text{function of } y$

Represent the variation of stress along y axis.

~~the variation of stress~~

(as the fiber of a beam.)

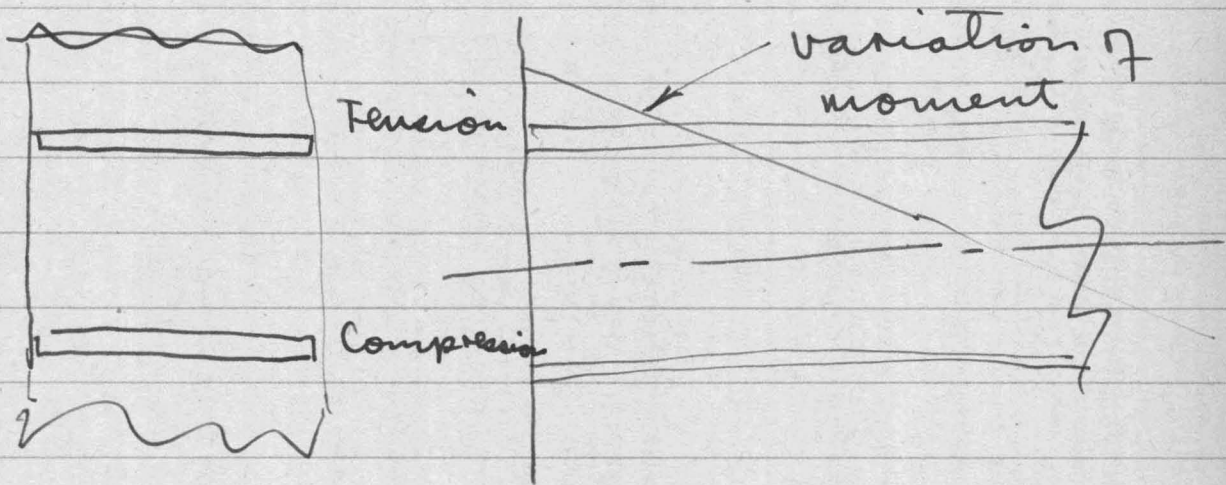


Actual solution

Worst case approach

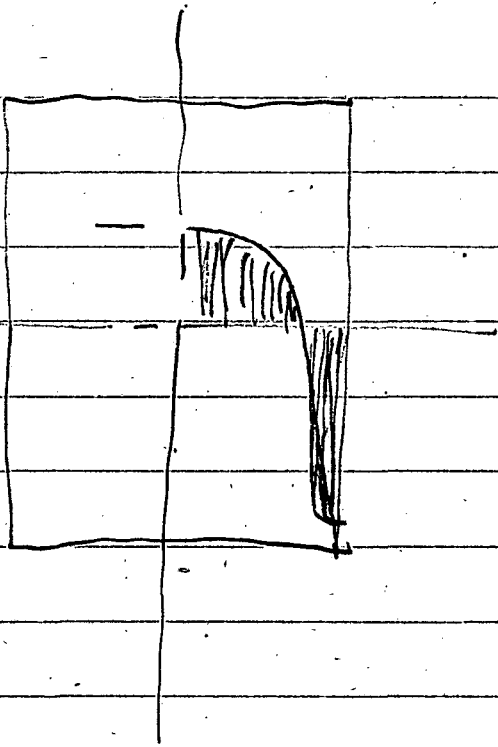
Average stress may be a closer <sup>(approach)</sup>

Take the joint as two welded plates.



Use the above method to find the relaxation  $\Delta L'_1$  &  $\Delta L'_2$

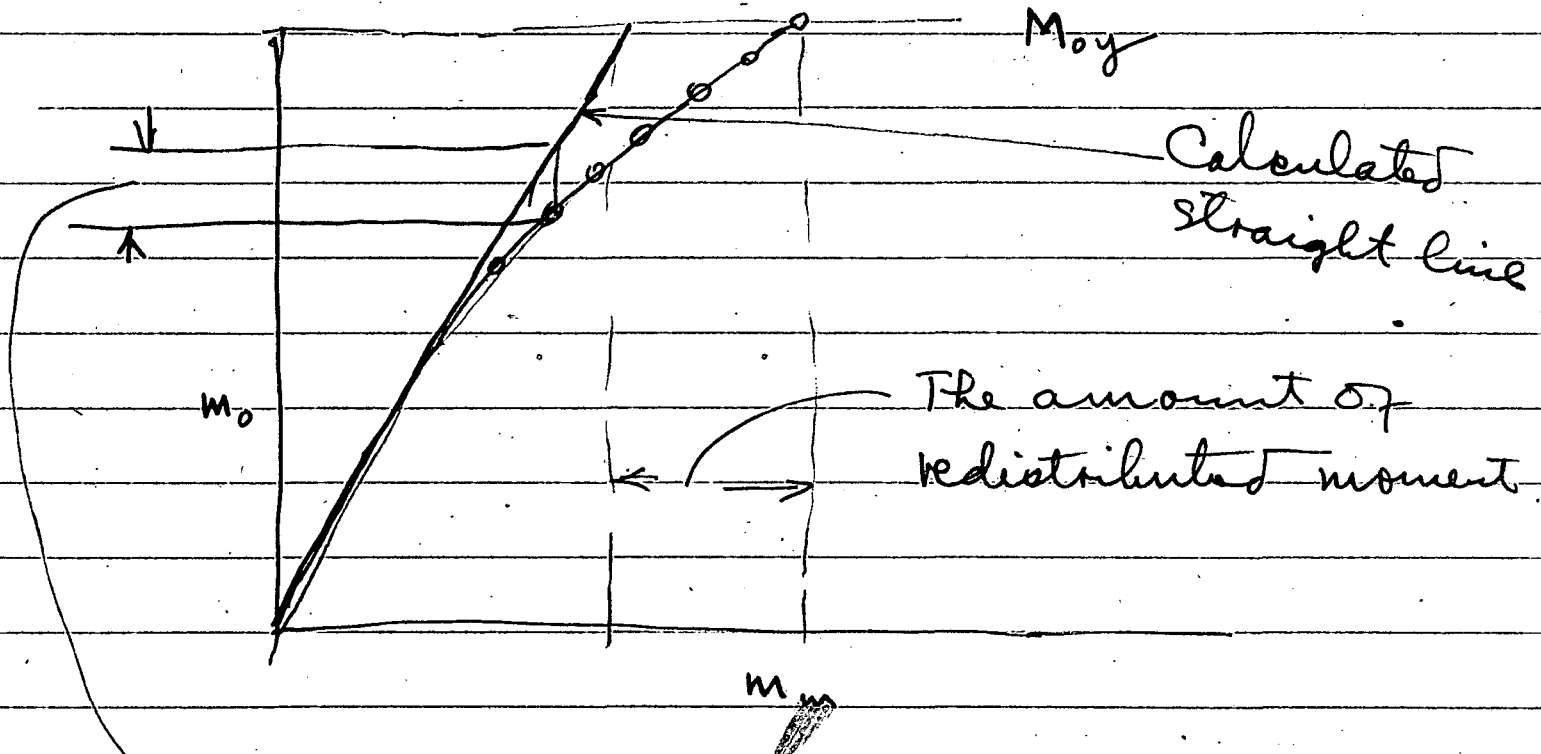
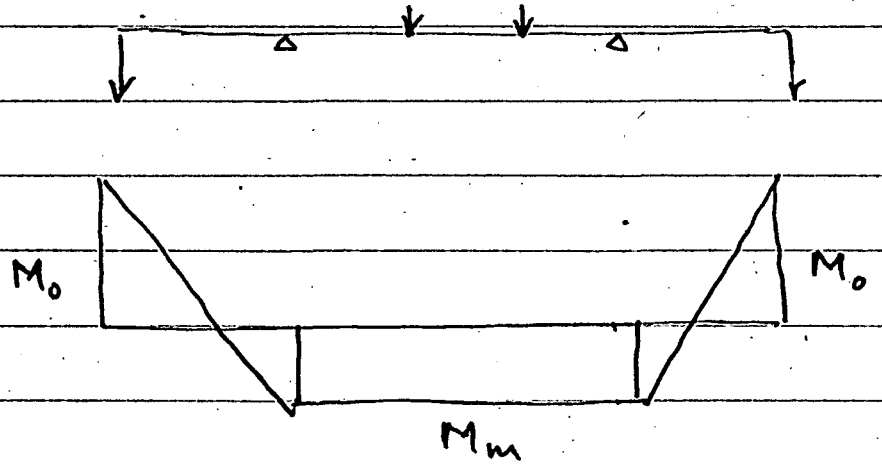
$$(\Delta L'_1 + \Delta L'_2) / h = \text{relaxation angle.}$$



~~Test Report~~

Test B2

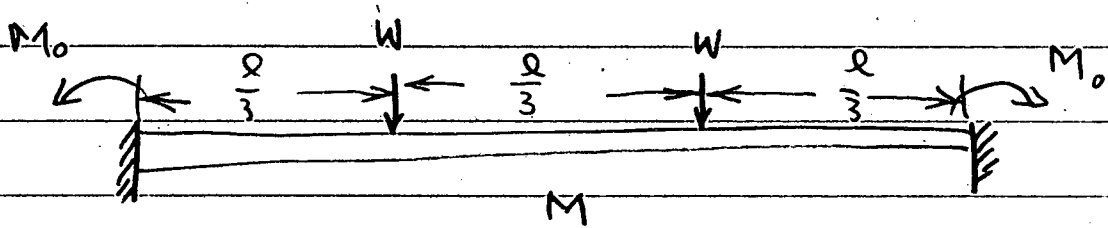
In the elastic range



The amount of moment reduced due to plastic relaxation.



Elastic behaviors of beams are affected by residual stresses as we have pointed out that the modulus of elasticity of the material is reduced by the presence of residual stresses.

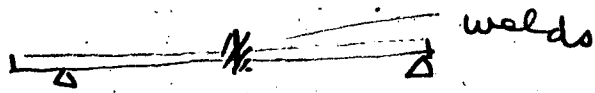


$$M_0 = F(W) = F(M) = \alpha M$$

In the case of a beam loaded as above we know

$$M_0 = 2M \quad \alpha = 2$$

1. Suppose only residual stresses due to rolling are considered and they are assumed to be uniform. Then the const.  $\alpha$  wouldn't be changed as load increases below the calculated yield strength.
2. a. Residual stresses only in central part. It is clear  $\alpha$  tends to increase.



8.

b. Even if there is no residual stress, ~~but~~  $\alpha$  is smaller than one. Central part of the beam reaches the plastic range first so reduces the rigidity of the central part of the beam. That will make

$\alpha$  tends to increase.

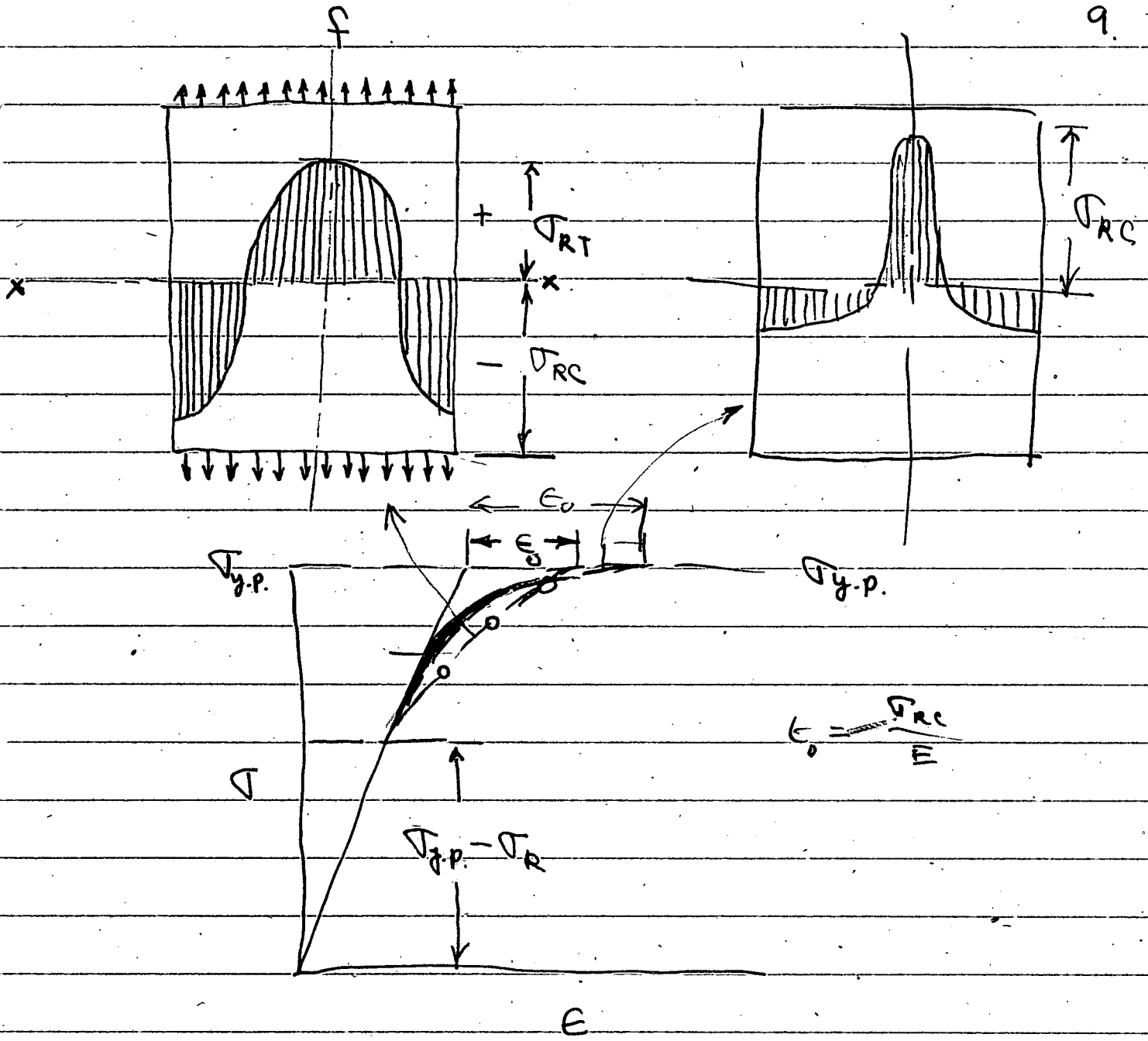
Maybe some tests should be done in case  $\alpha < 1$

3. Residual stresses are located at the ends of the beam. Then we will find

$\alpha$  tends to decrease.

See page 6.

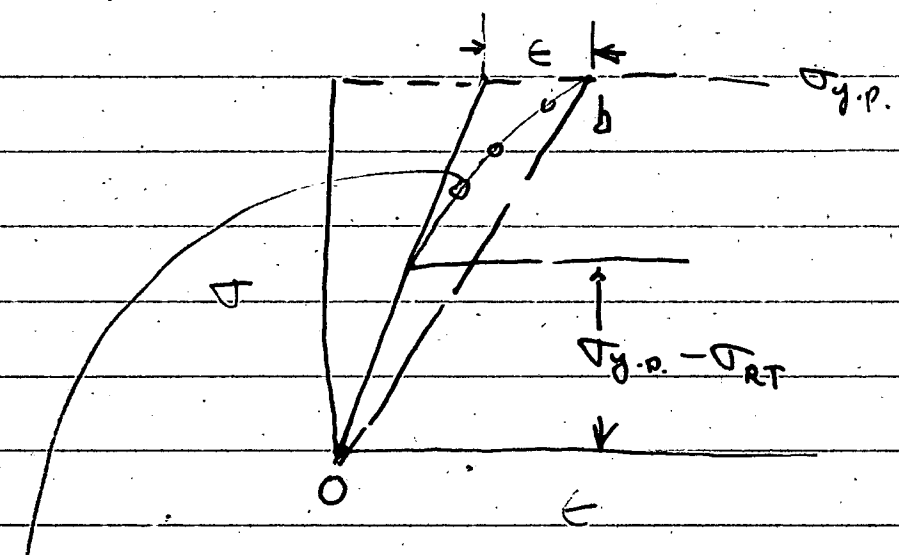
Therefore during a test we can always plot  $M$  vs  $M_0$  to find the value of  $\alpha$  and see how residual stresses affect the behavior of beams.



~~$$\epsilon_0 = \frac{1}{E} (\sigma_{RT} - \sigma_{RC}) - \frac{\sigma_{y.p.}}{E} = \frac{1}{E} (\sigma_{RT} - \sigma_{RC} - \sigma_{y.p.})$$~~

Where  $\sigma_{RT}$  = Mag of Tensile Residual stress

$\sigma_{RC}$  = Mag of Comp residual stress



This new straight line could be recommended in design. The corresponding Residual stress diagram is on page 5.

By using of equations on page 4 we can find the amount of relaxed moment.

→ This curve is very important for tangent modulus load of columns